Institution Explorer Post 1010

Milestone FRR Submittal

Vehicle Properties		
Total Length (in)	69.685	
Diameter (in)	4 in	
Gross Lift Off Weigh (lb)	7.45 lbs	
Airframe Material(s)	Thick walled paper tubes	
Fin Material and Thickness (in)	1/4 in Plywood	
Coupler Length(s)/Shoulder Length(s) (in)	Upper section shoulder length	

Motor Properties		
Motor Brand/Designation	Cesaroni J357-14A	
Max/Average Thrust (lb)	115.3 Lbf / 80.3 Lbf	
Total Impulse (lbf-s)	441.9 Lbfs	
Mass Before/After Burn (oz)	21.2oz / 9.3oz	
Liftoff Thrust (N)	422.6 N (95.0 Lbf)	
Motor Retention Method	Retention ring and screw	

Stability Analy	ysis et al.
Center of Pressure (in. from nose)	50.79
Center of Gravity (in. from nose)	40.16
Static Stability Margin (on pad)	2.58
Static Stability Margin (at rail exit)	2.69
Thrust-to-Weight Ratio	12.75:1
Rail Size/Type and Length (in)	1010, 96 in
Rail Exit Velocity (ft/s)	81.06

Ascent Analysis		
Maximum Velocity (ft/s)	573.917	
Maximum Mach Number	0.51	
Maximum Acceleration (ft/s^2)	564.304	
Target Apogee (ft)	3600	
Predicted Apogee (From Sim.) (ft)	3293.9	

Recovery System Properties - Overall	
Total Descent Time (s)	88
Total Drift in 20 mph winds (ft)	1974.25

Recovery System Properties - Energetics			
Ejection System Energetics (ex. Black Powder)		Black powder	
Energetics Mass - Drogue	Primary	0.4	
Chute (grams)	Backup	0.5	
Energetics Mass - Main Chute	Primary	1	
(grams)	Backup	1	
Energetics Mass - Other	Primary	N/A	
(grams) - If Applicable	Backup		

Recovery System Pr	operties - R	ecovery Electronics	
Primary Altimeter Make	e/Model	RRC3 "Sport" Altimeter	
Secondary Altimeter Mal	ke/Model	RRC3 "Sport" Altimeter	
Other Altimeters (if app	olicable)	N/A	
Rocket Locator (Make/	Model)	Featherweight Tracker	
Additional Locators (if applicable)		N/A	
Transmitting Frequencies (all payload)	- vehicle and	2450 MHz, 921 MHz	
Describe Redundancy Plan (batteries, switches, etc.)	separate systems and batteries, backup altimeter set to deploy 1 second after apogee; Two redundant ejection charge wells for drogue and two separate		
Pad Stay Time (Launch Configuration)	2 Hours		

Recovery System Properties - Drogue Parachute					
Ma	nufacturer/Mo	del	Fruity Chutes		
Size o	or Diameter (in	or ft)	12"		
Main Altin	neter Deployme	ent Setting	At apogee		
Backup Alti	meter Deploym	ent Setting	1 second af	ter apogee	
Velocit	y at Deploymer	nt (ft/s)	0 f	t/s	
Terminal Velocity (ft/s)		t/s)	48 ft/s		
Recovery Harness Material, Size, and Type (examples - 1/2 in. tubular Nylon or 1 in. flat Kevlar strap)		1000 lb rated kevlar line			
Recovery Harness Length (ft)		gth (ft)	20 ft		
Harness/Airframe Interfaces		Eye bolt on ebay			
Kinetic	Section (Ebay)	on 2 (Lower sec	Section 3	Section 4	
Energy of Each Section (Ft-lbs)	88.4	84.7	N/A	N/A	

Recovery System Properties - Main Parachute					
Ma	nufacturer/Mo	del	Fruity chutes		
Size o	or Diameter (in	or ft)	36 "		
Main Altime	ter Deploymen	t Setting (ft)	600 ft		
Backup Altim	eter Deployme	nt Setting (ft)	500) ft	
Velocit	y at Deploymer	nt (ft/s)	48 1	ft/s	
Terminal Velocity (ft/s)		18 ft/s			
Recovery Harness Material, Size, and Type (examples - 1/2 in. tubular Nylon or 1 in. flat Kevlar strap)		1000 lb rated kevlar line			
Recove	ery Harness Len	arness Length (ft) 15 ft			
Harness/Airframe Interfaces E		Eye bolt on ebay			
Kinetic	ction 1 (Payloa	on 1 (Payloa ection 2 (Ebay on 3 (Lower sec Section			
Energy of Each Section (Ft-lbs)	20.11	12.4	11.9	N/A	

		_		
Institution	Explorer Post 1010		Milestone	FRR Submittal

	Payload
	Overview
Payload 1 (official payload)	Autonomous Guided Recovery System - The payload will separate from the rocket at apogee. At 500 ft, two redundant jolly logic chute releases will allow a guided parafoil to deploy. At 400 ft the parafoil will autonomously guide itself back to the launch site.
	Overview
Payload 2 (non-scored payload)	N/A

	Test Plans, Status, and Results (WILL BE UPDATED WITH STATUS AND RESULTS)
Ejection Charge Tests	Subscale Rocket: We performed two ground tests of the ejection system, with 0.36 grams of black powder forward and 0.48 g aft. Both separations were successful, although we determined that less black powder should be used than initially thought. Additionally, the shear pins slightly zippered the tube so the paper tubes will have to be reinforced more.
Sub-scale Test Flights	The rocket successfully reached an apogee of 600 feet, which was considerably under our target of 800ft. The rocket was 40 grams heavier than what was predicted in OpenRocket, due to some small differences in hardware used. Total flight time was 46 seconds.
Vehicle Demonstratio n Flights	On February 12 the full scale rocket successfully was launched and recovered. According to the featherweight GPS, the vehicle reached an apogee of 3472 ft. The RRC3 recorded an apogee of 3369 ft. The flight was conducted with a half-kilogram mass simulator in the nosecone.
Payload Demonstratio n Flights	Plan to launch in March (12/13). This test launch's conditions will be as close as possible to the competition launch.

				_
Institution	Explorer Post 1010	Milest	estone FRR Submittal	

	Transmitte	er #1		
Location of transmitter:	Nosecone			
Purpose of transmitter:	GPS tracking to aid in recovery			
Brand	Nordic Semiconductor	RF Output Power (mW)	80 mW	
Model	NRF24I01+	Specific Frequency used by team (MHz)	2450	
Handshake or frequency hopping? (explain)	N/A			
Distance to closest e-match or altimeter (in)	3 in			
Description of shielding plan:	Separated by plywood bulkhead			

Transmitter #2				
Location of transmitter:	Electronics bay			
Purpose of transmitter:	GPS tracking transmission to aid in recovery			
Brand	Featherweight Altimeters	RF Output Power (mW)	250 mW	
Model	Featherwight GPS Tracker	Specific Frequency used by team (MHz)	921	
Handshake or frequency hopping? (explain)				
Distance to closest e-match or altimeter (in)	~1.5 in			
Description of shielding plan:	Separated by plywood bulkhead			

Transmitter #3			
Location of transmitter:	N/A		
Purpose of transmitter:			
Brand	RF Output Power (mW)		
Model	Specific Frequency used by team (MHz)		
Handshake or frequency hopping? (explain)			
Distance to closest e-match or altimeter (in)			
Description of shielding plan:			

Transmitter #4			
Location of transmitter:	N/A		
Purpose of transmitter:			
Brand	RF Output Power (mW)		
Model	Specific Frequency used by team (MHz)		
Handshake or frequency hopping? (explain)			
Distance to closest e-match or altimeter (in)			
Description of shielding plan:			

Institution	Explo	orer Post 1010	Milestone	FRR Submittal		
			-			
Transmitter #5						
Location	of transmitter:		N/A			
Purpose	of transmitter:					
Brand			RF Output Power (mW	/)		
	Model		Specific Frequency used by tea	am (MHz)		
Handshake or freq	uency hopping? (explain)					
Distance to closest	t e-match or altimeter (in)					
Description	n of shielding plan:					
		Transmitter #6	5			
	of transmitter:		N/A			
Purpose	of transmitter:					
	Brand		RF Output Power (mW	/)		
	Model		Specific Frequency used by tea	am (MHz)		
Handshake or freq	uency hopping? (explain)					
Distance to closest	t e-match or altimeter (in)					
Description	n of shielding plan:					
		Additional Comm	ents			