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Explorer Post 1010

Vehicle Properties		
Total Length (in)	64.56 in	
Diameter (in)	4 in	
Gross Lift Off Weigh (lb)	5.32	
Airframe Material(s)	Thick walled paper tubes	
Fin Material and Thickness (in)	1/4 in Plywood	
Coupler Length(s)/Shoulder Length(s) (in)	Upper section shoulder length	

Motor Properties		
Motor Brand/Designation	Cesaroni J357-14A	
Max/Average Thrust (lb)	115.3 Lbf / 80.3 Lbf	
Total Impulse (Ibf-s)	441.9 Lbfs	
Mass Before/After Burn (oz)	21.2oz / 9.3oz	
Liftoff Thrust (N)	422.6 N (95.0 Lbf)	
Motor Retention Method	Retention ring and screw	

Stability Analysis		
Center of Pressure (in. from nose)	46.063 in	
Center of Gravity (in. from nose)	35.472 in	
Static Stability Margin (on pad)	2.65	
Static Stability Margin (at rail exit)	2.698	
Thrust-to-Weight Ratio	18.26:1	
Rail Size/Type and Length (in)	1010, 96 in	
Rail Exit Velocity (ft/s)	98.02 ft/s	

Ascent Analysis		
Maximum Velocity (ft/s)	757.87	
Maximum Mach Number	0.68	
Maximum Acceleration (ft/s^2)	659.45	
Target Apogee (ft)	3750	
Predicted Apogee (From Sim.) (ft)	3677.82	

Recovery System Properties - Overall		
Total Descent Time (s)	87.33	
Total Drift in 20 mph winds (ft)	1560.03	

Recovery System Properties - Energetics			
Ejection System Energetics (ex. Black Powder)		Black powder	
Energetics Mass - Drogue	Primary	1.5	
Chute (grams)	Backup	2.1	
Energetics Mass - Main Chute (grams)	Primary	1.5	
	Backup	2.1	
Energetics Mass - Other (grams) - If Applicable	Primary	N/A	
	Backup		

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Recovery System Properties - Recovery Electronics			
Primary Altimeter Make	e/Model	RRC3 "Sport" Altimeter	
Secondary Altimeter Mal	ke/Model	RRC3 "Sport" Altimeter	
Other Altimeters (if app	licable)	N/A	
Rocket Locator (Make/	Model)	Featherweight Tracker	
Additional Locators (if applicable)		N/A	
Transmitting Frequencies (all - vehicle and payload)		915 MHz, 921 MHz	
Describe Redundancy Plan (batteries, switches, etc.)	Two redundant altimeters with completely separate systems and batteries, backup altimeter set to deploy 1 second after apogee; Two redundant ejection charge		
Pad Stay Time (Launch Configuration)	2 Hours		

Recovery System Properties - Drogue Parachute				
Manufacturer/Model		Fruity Chutes		
Size c	or Diameter (in	or ft)	12"	
Main Altin	neter Deployme	ent Setting	At apogee	
Backup Altimeter Deployment Setting		1 second after apogee		
Velocity at Deployment (ft/s)		0 ft/s		
Terminal Velocity (ft/s)		66.27 ft/s		
Recovery Harr (examples - 1 f	Recovery Harness Material, Size, and Type (examples - 1/2 in. tubular Nylon or 1 in. flat Kevlar strap)		1000 lb rated kevlar line	
Recove	ry Harness Len	ngth (ft) 20 ft		ft
Harness/Airframe Interfaces		Eye bolt on ebay		1
Kinetic Energy of Each Section (Ft-lbs)	Section 1	Section 2	Section 3	Section 4
	98.22	192.08	N/A	N/A

Recovery System Properties - Main Parachute				
Ma	inufacturer/Mo	del	Fruity chutes	
Size o	or Diameter (in o	or ft)	36 "	
Main Altime	ter Deployment	t Setting (ft)	800 ft	
Backup Altim	eter Deploymer	nt Setting (ft)	700 ft	
Velocity at Deployment (ft/s)		64.30) ft/s	
Terminal Velocity (ft/s)		18.4 ft/s		
Recovery Harness Material, Size, and Type (examples - 1/2 in. tubular Nylon or 1 in. flat Kevlar strap)		1000 lb rated kevlar line		
Recovery Harness Length (ft) 15 ft			ft	
Harness/Airframe Interfaces		Eye bolt on ebay		
Kinetic	ction 1 (Payloa	Section 2	Section 3	Section 4
Energy of Each Section (Ft-lbs)	12.86	9.65	18.88	N/A

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	Payload	
	Overvie	2W
Payload 1 (official payload)	Autonomous Guided	Recovery System
	Overvie	2W
Payload 2 (non-scored payload)	N/A	

Test Plans, Status, and Results (WILL BE UPDATED WITH STATUS AND RESULTS)			
Ejection Charge Tests	Plan to test ejection charges by carrying out separation tests on the ground.		
Sub-scale Test Flights	Sub-scale rocket test flight planned for late December.		
Vehicle Demonstratio n Flights	Vehicle demonstration flight planned for Februrary, with the payload recovered via unguided parachute.		
Payload Demonstratio n Flights	Plan to launch in March. This test launch's conditions will be as close as possible to the competition launch.		

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Transmitter #1				
Location of transmitter:	Nosecone			
Purpose of transmitter:	GPS tracking to aid in recovery			
Brand	Featherweight	RF Output Power (mW)		
Model	GPS Tracker	Specific Frequency used by team (MHz)	921	
Handshake or frequency hopping? (explain)				
Distance to closest e-match or altimeter (in)	3 in			
Description of shielding plan:	Separated by plywood bulkhead			

Transmitter #2				
Location of transmitter:	Electronics bay			
Purpose of transmitter:	GPS tracking transmission to aid in recovery			
Brand	Adafruit	RF Output Power (mW)		
Model	Adafruit LoRa Featherweight RFM95W	Specific Frequency used by team (MHz)	915	
Handshake or frequency hopping? (explain)				
Distance to closest e-match or altimeter (in)	2.5 in			
Description of shielding plan:	Separated by plywood bulkhead			

Transmitter #3		
Location of transmitter:	N/A	
Purpose of transmitter:		
Brand	RF Output Power (mW)	
Model	Specific Frequency used by team (MHz)	
Handshake or frequency hopping? (explain)		
Distance to closest e-match or altimeter (in)		
Description of shielding plan:		

Transmitter #4		
Location of transmitter:	N/A	
Purpose of transmitter:		
Brand	RF Output Power (mW)	
Model	Specific Frequency used by team (MHz)	
Handshake or frequency hopping? (explain)		
Distance to closest e-match or altimeter (in)		
Description of shielding plan:		

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Transmitter #5		
Location of transmitter:	N/A	
Purpose of transmitter:		
Brand	RF Output Power (mW)	
Model	Specific Frequency used by team (MHz)	
Handshake or frequency hopping? (explain)		
Distance to closest e-match or altimeter (in)		
Description of shielding plan:		

Transmitter #6		
Location of transmitter:	N/A	
Purpose of transmitter:		
Brand	RF Output Power (mW)	
Model	Specific Frequency used by team (MHz)	
Handshake or frequency hopping? (explain)		
Distance to closest e-match or altimeter (in)		
Description of shielding plan:		

Additional Comments